NUMERICAL STUDIES OF PROTOTYPE CAVITY FLOW PROBLEMS

by Donald Greenspan

APPENDIX: CDC 3600 FORTRAN PROGRAM FOR CAVITY FLOW PROBLEMS

by D. Schultz

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Donald Greenspan

1. Introduction.

The flow of a gas or of a liquid in a closed cavity has long been of interest in applied science (see, e.g., references [1,2,4,7-12,14] and the additional references contained therein). In this paper we will apply the power of the high speed digital computer to study prototype, steady state, two dimensional problems for such flows. The numerical methods to be developed will be finite difference methods and will be described in sufficient generality so as to be applicable to nonlinear coupled systems similar in structure to the Navier-Stokes equations.

2. The Eddy Problem in a Rectangle.

The class of problems to be studied, called eddy problems in a rectangle, can be formulated as follows. For d > 0, let the points (0,0), (1,0), (1,d) and (0,d) be denoted by A, B, C and D, respectively (see Figure 2.1). Let S be the rectangle whose vertices are A, B, C, D and denote its interior by R. On R the equations of motion to be satisfied are the Navier-Stokes equations, that is

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$$\Delta \psi = -\omega$$

(2.2)
$$\Delta \omega + \mathcal{B} \left(\frac{\partial x}{\partial \psi} \frac{\partial y}{\partial \omega} - \frac{\partial y}{\partial \psi} \frac{\partial x}{\partial \omega} \right) = 0 ,$$

where ψ is the stream function, ω is the vorticity and \Re is the Reynolds number. On S the boundary conditions to be satisfied are

$$(2.3) \psi = 0, \frac{\partial \psi}{\partial x} = 0 , \text{ on AD}$$

(2.4)
$$\psi = 0, \quad \frac{\partial \psi}{\partial y} = 0 \quad , \quad \text{on AB}$$

(2.5)
$$\psi = 0, \frac{\partial \psi}{\partial \mathbf{x}} = 0$$
 , on BC

(2.6)
$$\psi = 0$$
, $\frac{\partial \psi}{\partial y} = -1$, on CD.

The analytical problem is defined on R+S by (2.1)-(2.6) and is shown diagrammatically in Figure 2.1.

In general, boundary value problem (2.1)-(2.6) cannot be solved by means of existing analytical techniques. Physical solutions have been produced in the laboratory by Pan and Acrivos [9], while numerical methods which "converge", but only for small \Re , have been developed by Burggraf [4] and Runchal, Spalding and Wolfshtein [12]. A numerical method which converges for all \Re , but which has been run only for relatively large values of the grid size, has been developed by the writer [7].

We shall describe next a modified, somewhat faster form of the method developed in [7] and apply it to a selection of difficult problems which are

of wide interest. Among our major objectives will be the construction of secondary vortices and the study of vorticity for large Reynolds number.

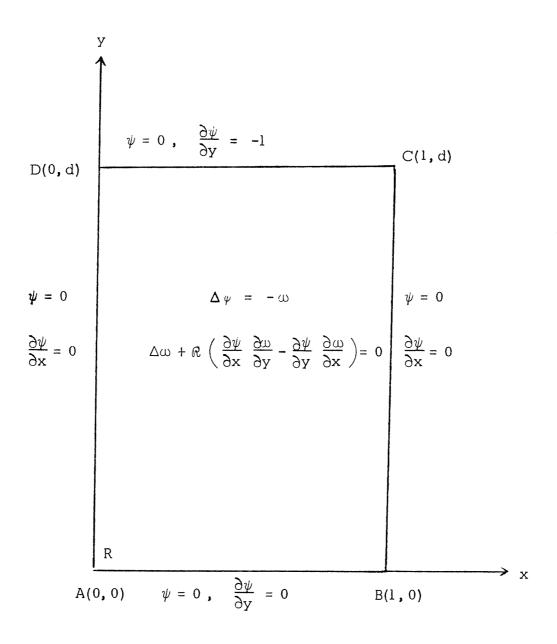


Figure 2.1

3. The General Numerical Method.

For a fixed positive integer n, set $h=\frac{1}{n}$. Assume, for simplicity, that d is an integral multiple of h. (If d is not an integral multiple of h, the method is easily modified as shown in [7].) Starting at (0,0) with grid size h, construct and number in the usual way [7] the set of interior grid points R_h and the set of boundary grid points S_h .

For given tolerances $~\epsilon_1~$ and $~\epsilon_2,~$ we will show first how to construct on R_h a sequence of discrete stream functions

(3.1)
$$\psi^{(0)}, \psi^{(1)}, \psi^{(2)}, \cdots$$

and on $R_h + S_h$ a sequence of discrete vorticity functions

(3.2)
$$\omega^{(0)}, \omega^{(1)}, \omega^{(2)}, \cdots,$$

such that for some integer k both the following are valid:

$$|\psi^{(k)} - \psi^{(k+1)}| < \epsilon_1, \quad \text{on } R_h$$

(3.4)
$$|\omega^{(k)} - \omega^{(k+1)}| \leq \varepsilon_2, \quad \text{on } R_h + S_h.$$

Initially, set

(3.5)
$$\psi^{(0)} = C_1$$
 , on R_h

(3.6)
$$\omega^{(0)} = C_2$$
, on $R_h + S_h$,

where C_1 and C_2 are constants.

To produce the second iterate $\psi^{(1)}$ of sequence (3.1) proceed as follows. At each point of R_h of the form (h,ih), $i=2,\cdots,n-2$, approximate (2.3) by

(3.7)
$$\psi(h, ih) = \frac{\psi(2h, ih)}{4}$$
.

At each point of R_h of the form (ih,h), i = 1,2,...,n-1, approximate (2.4) by

(3.8)
$$\psi(ih, h) = \frac{\psi(ih, 2h)}{4}$$
.

At each point of R_h of the form (1-h,ih), $i=2,3,\cdots,n-2$, approximate (2.5) by

(3.9)
$$\psi(1-h, ih) = \frac{\psi(1-2h, ih)}{4}.$$

At each point of R_h of the form (ih, 1-h), i = 1, 2, ..., n-1, approximate (2.6) by

(3.10)
$$\psi(ih, l-h) = \frac{h}{2} + \frac{\psi(ih, l-2h)}{4}.$$

And at each remaining point of $R_{\mbox{\scriptsize h}}$ write down the difference analogue

(3.11)
$$-4\psi(x,y) + \psi(x+h,y) + \psi(x,y+h) + \psi(x-h,y) + \psi(x,y-h) = -h^2 \omega^{(0)}(x,y)$$

of (2.1). Solve the linear algebraic system generated by (3.7)-(3.11) by the generalized Newton's method [7] with over-relaxation factor \mathbf{r}_{ψ} and denote this solution by $\overline{\psi}^{(1)}$. Then, on \mathbf{R}_{h} , $\psi^{(1)}$ is defined by the smoothing formula

(3.12)
$$\psi^{(1)} = \rho \psi^{(0)} + (1-\rho) \overline{\psi}^{(1)}, \quad 0 \le \rho \le 1.$$

To produce the second iterate $\omega^{(1)}$ of sequence (3.2) proceed as follows. At each point of S_h of the form (ih,0), i = 0,1,2,...,n, set

(3.13)
$$\overline{\omega}^{(1)}(ih, 0) = -\frac{2\psi^{(1)}(ih, h)}{h^2};$$

at each point of S_h of the form (0, i h), i = 1, 2, ..., n-1, set

(3.14)
$$\overline{\omega}^{(1)}(0, ih) = -\frac{2\psi^{(1)}(h, ih)}{h^2};$$

at each point of S_h of the form (l, ih), i = l, 2, ..., n-l, set

(3.15)
$$\overline{\omega}^{(1)}(1, ih) = -\frac{2\psi^{(1)}(1-h, ih)}{h^2};$$

and at each point of S_h of the form (ih,1), i = 0,1,2,...,n, set

(3.16)
$$\overline{\omega}^{(1)}(ih, 1) = \frac{2}{h} - \frac{2\psi^{(1)}(ih, 1-h)}{h^2}$$
.

Next, at each point (x,y) in R_h set

$$\alpha = \psi^{(1)}(x+h, y) - \psi^{(1)}(x-h, y)$$

$$\beta = \psi^{(1)}(x, y+h) - \psi^{(1)}(x, y-h)$$

and approximate (2.2), appropriately, by

(3.17)
$$(-4 - \frac{\alpha \Re}{2} - \frac{\beta \Re}{2}) \omega(x, y) + \omega(x+h, y) + (1 + \frac{\alpha \Re}{2}) \omega(x, y+h)$$

$$+ (1 + \frac{\beta \Re}{2}) \omega(x-h, y) + \omega(x, y-h) = 0; \text{ if } \alpha \ge 0, \beta \ge 0,$$

(3.18)
$$(-4 - \frac{\alpha_{\Re}}{2} + \frac{\beta_{\Re}}{2})\omega(x, y) + (1 - \frac{\beta_{\Re}}{2})\omega(x+h, y) + (1 + \frac{\alpha_{\Re}}{2})\omega(x, y+h)$$
$$+ \omega(x-h, y) + \omega(x, y-h) = 0 ; \quad \text{if } \alpha \ge 0, \quad \beta < 0 ,$$

$$(3.19) \qquad (-4 + \frac{\alpha \Re}{2} - \frac{\beta \Re}{2}) \omega(x, y) + \omega(x+h, y) + \omega(x, y+h) + (1 + \frac{\beta \Re}{2}) \omega(x-h, y)$$

$$+ (1 - \frac{\alpha \Re}{2}) \omega(x, y-h) = 0 ; \quad \text{if } \alpha < 0, \quad \beta \ge 0 ,$$

(3.20)
$$(-4 + \frac{\alpha \Re}{2} + \frac{\beta \Re}{2}) \omega(x, y) + (1 - \frac{\beta \Re}{2}) \omega(x+h, y) + \omega(x, y+h)$$
$$+ \omega(x-h, y) + (1 - \frac{\alpha \Re}{2}) \omega(x, y-h) = 0 ; \text{ if } \alpha < 0, \beta < 0 .$$

Solve the linear algebraic system generated by (3.17)-(3.20) by the generalized Newton's method with over-relaxation factor \mathbf{r}_{ω} and denote the solution by $\overline{\omega}^{(1)}$. Finally, on all of $\mathbf{R}_h + \mathbf{S}_h$ define $\omega^{(1)}$ by the smoothing formula

$$\omega^{(1)} = \mu \omega^{(0)} + (1 - \mu) \overline{\omega}^{(1)}$$
, $0 \le \mu \le 1$.

Proceed next to determine $\psi^{(2)}$ on R_h from $\omega^{(1)}$ and $\psi^{(1)}$ in the same fashion as $\psi^{(1)}$ was determined from $\omega^{(0)}$ and $\psi^{(0)}$. Then construct $\omega^{(2)}$ on $R_h + S_h$ from $\omega^{(1)}$ and $\psi^{(2)}$ in the same fashion as $\omega^{(1)}$ was determined from $\omega^{(0)}$ and $\psi^{(1)}$. In the indicated fashion, construct the sequences (3.1) and (3.2). Terminate the computation when (3.3) and (3.4) are valid.

Finally, when $\psi^{(k)}$ and $\omega^{(k)}$ are verified to be solutions of the difference analogues of (2.1) and (2.2), they are taken to be the numerical approximations of $\psi(x,y)$ and $\omega(x,y)$, respectively.

4. Examples.

Consider first the boundary value problem defined by (2.1)-(2.6) with d=1. This problem was solved by the method of Section 3 for $\Re=200$ with $h=\frac{1}{20}$, $\epsilon_1=1$, $\epsilon_2=10^{-4}$, $\rho=0.1$, $\mu=0.7$, $r_{\psi}=1.8$, $r_{\omega}=1.0$, $C_1=C_2=0$, and also for $\Re=500$, 2000 and 15000 with the same parameter values except for $\epsilon_2=10^{-3}$. Convergence was achieved for $\Re=200$ in 14 minutes with 341 outer iterations, for $\Re=500$ in 11 minutes with 96 outer iterations, for $\Re=2000$ in 4 minutes with 80 outer iterations, and for $\Re=15000$ in $3\frac{1}{2}$ minutes with 40 outer iterations. The resulting stream curves exhibited only primary vortices and are shown in Figure 4.1. The resulting equivorticity curves exhibited the double spiral development shown in [7] and are given in Figure 4.2.

With an aim toward producing secondary vortices and toward studying vorticity for large Reynolds numbers, boundary value problems (2.1)-(2.6) was considered again with d = 1. The problem was solved for \Re = 50, 10000, and 100000 with h = $\frac{1}{40}$. For \Re = 50 the remaining input parameters were chosen to be ε_1 = 10^{-4} , ε_2 = 10^{-3} , ρ = .03, μ = .90, r_{ψ} = 1.8, r_{ω} = 1.8, r_{ψ} = 1.8, r_{ψ}

single secondary vortex is shown in Figure 4.4. For \Re = 100000, the remaining input parameters were chosen to be ϵ_1 = 10 $^{-4}$, ϵ_2 = .005, ρ = .03, μ = .95, r_{ψ} = 1.8, r_{ω} = 1, but $\psi^{(0)}$ and $\omega^{(0)}$ were taken to be the 57 th outer iterates of the run for \Re = 10000. Convergence was achieved in 135 minutes with 386 outer iterations. The flow is shown in Figure 4.5 and contains no secondary vortices. The equivorticity curve ω = 1.630, with its doublespiral, space filling characteristics is shown in Figure 4.6. Numerical evidence of Batchelor's result that the vorticity in a large subregion of R converges to a constant as $R \rightarrow \infty$ is exhibited in Figure 4.6 by setting crosses on those points at which the vorticity is between 1.6 and 1.7.

Finally, consider boundary value problem (2.1)-(2.6) with d = 2 and R = 10. This problem was solved with h = $\frac{1}{40}$, $\epsilon_1 = 10^{-4}$, $\epsilon_2 = 10^{-3}$, $\rho = .05$, $\mu = .85$, $r_{\psi} = 1.8$, $r_{\omega} = 1.25$, $C_1 = C_2 = 0$. Convergence was achieved in 32 minutes with 102 outer iterations. The resulting flow, with its two primary and two secondary vortices, is shown in Figure 4.7.

5. Remarks.

From the many examples run in addition to those described in Section 4, the following observations and heuristic conclusions resulted. Divergence or exceptionally slow convergence usually followed if any one of the following choices were made: $.4 \le \rho \le 1$, $0 \le \mu \le .6$, $r_{\psi} \le 1$, $r_{\psi} \le 1$. The choice $\rho = \mu = 0$ yields convergence only for large grid sizes and small Reynolds numerous

bers. The choice r_{ψ} = 1.8 was consistently good. For grid sizes larger than or equal to $\frac{1}{20}$, sequence (3.1) converged so much faster than (3.2) that very little attention had to be directed toward the choice of ϵ_1 , but for grids—smaller than $\frac{1}{20}$ this was not the case—and attention had to be directed to the choices of both ϵ_1 and ϵ_2 . Deletion of all or even of some of the special formulas (3.7)-(3.10) and substitution with (3.11) always led to divergence for large Reynolds numbers ($\Re \sim 10000$), but often did yield secondary vortices for $h = \frac{1}{20}$ for small Reynolds numbers ($\Re \sim 50$). The difference equations for $\psi^{(k)}$ and $\omega^{(k)}$ were always satisfied to much smaller tolerances than those imposed in (3.3) and (3.4), respectively.

Several possible modifications of the method of this paper which should be explored if one wishes to speed up the convergence include allowing some or all of ρ , μ , r_{ψ} and r_{ω} to be variable [6], using line over relaxation [15], and choosing $\psi^{(0)}$ and $\omega^{(0)}$ in a more judicious manner than that prescribed in (3.5)-(3.6).

Observe also that the method of Section 3 applies directly to biharmonic problems (i.e., to the case \Re = 0) and initial computations verify that it extends in a natural way to free convection problems [1].

Finally, note that theoretical support for the method of this paper is now beginning to appear for very special cases [3,5,13].

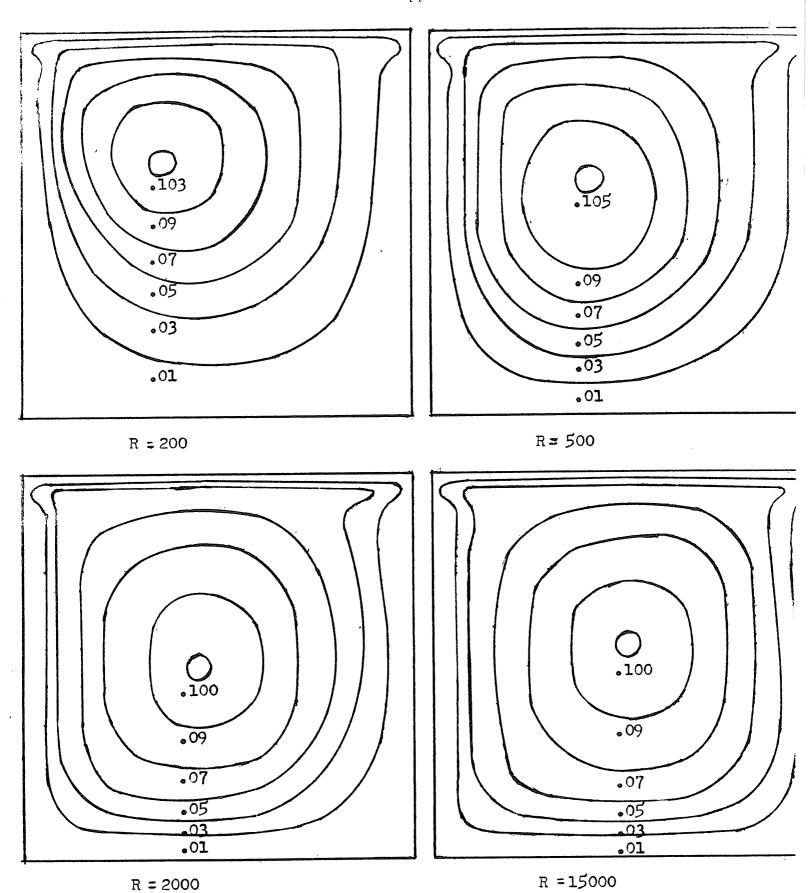


FIGURE 4.1 Typical streamlines for h = 1/20.

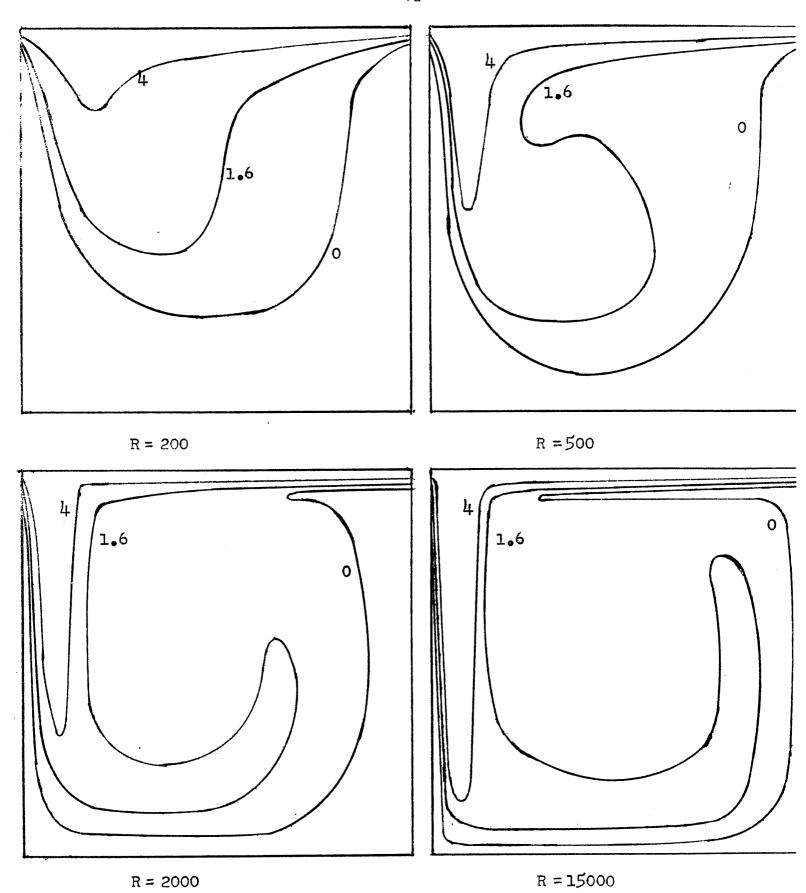


FIGURE 4.2 Selected equivorticity curves for h = 1/20.

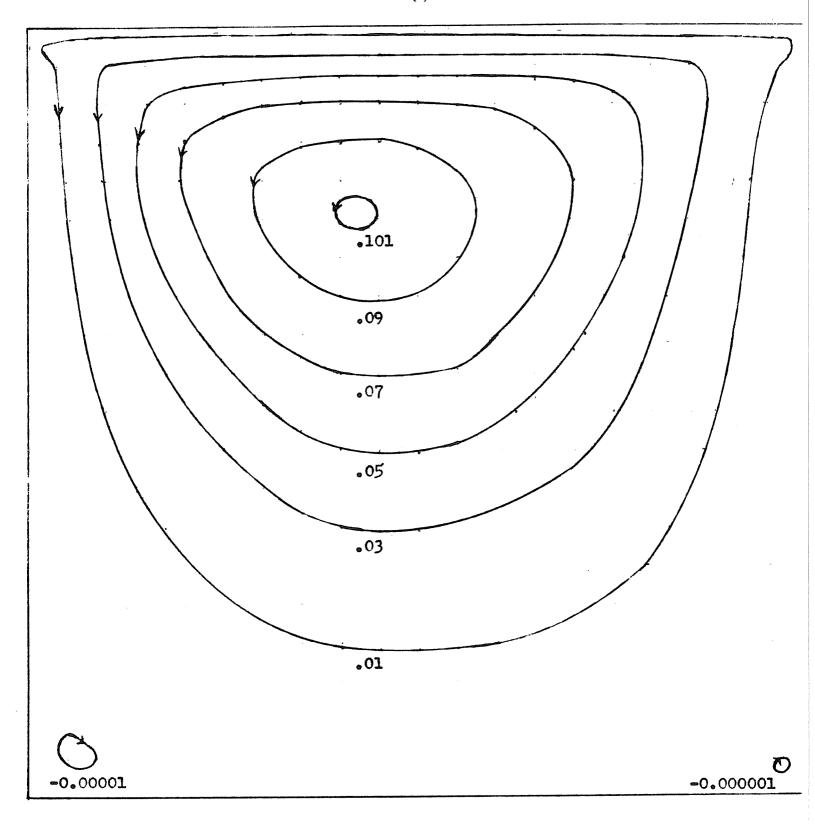


FIGURE 4.3 Streamlines for Reynolds number 50 with h = 1/40.

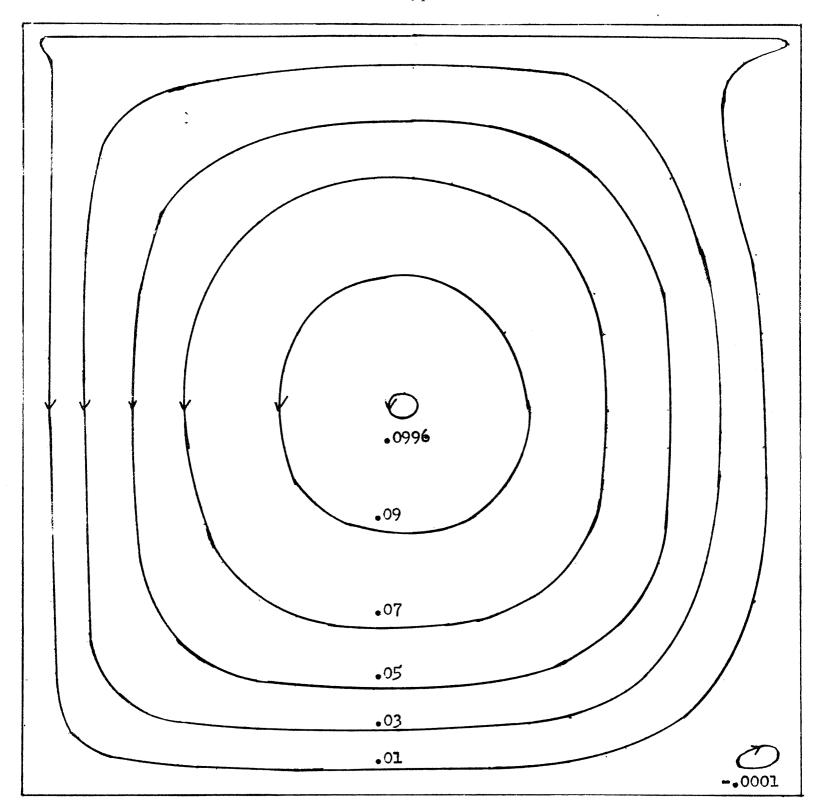


FIGURE $\mu_{\bullet}\mu_{\bullet}$ Streamlines for Reynolds number 10000 with $h=1/\mu_{0}$.

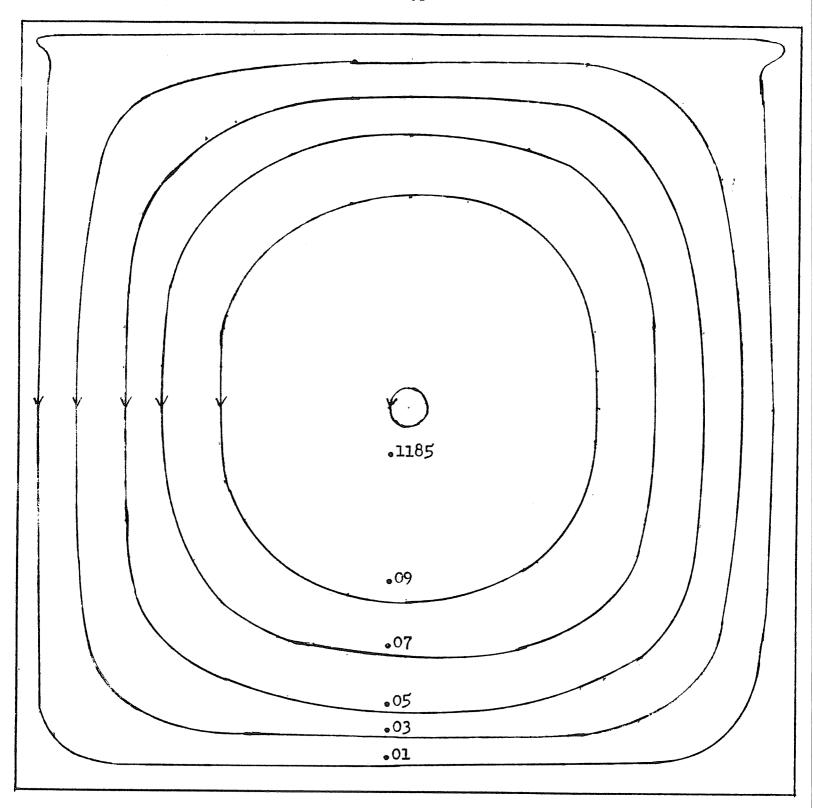


FIGURE 4.5 Streamlines for Reynolds number 100000 with h = 1/40.

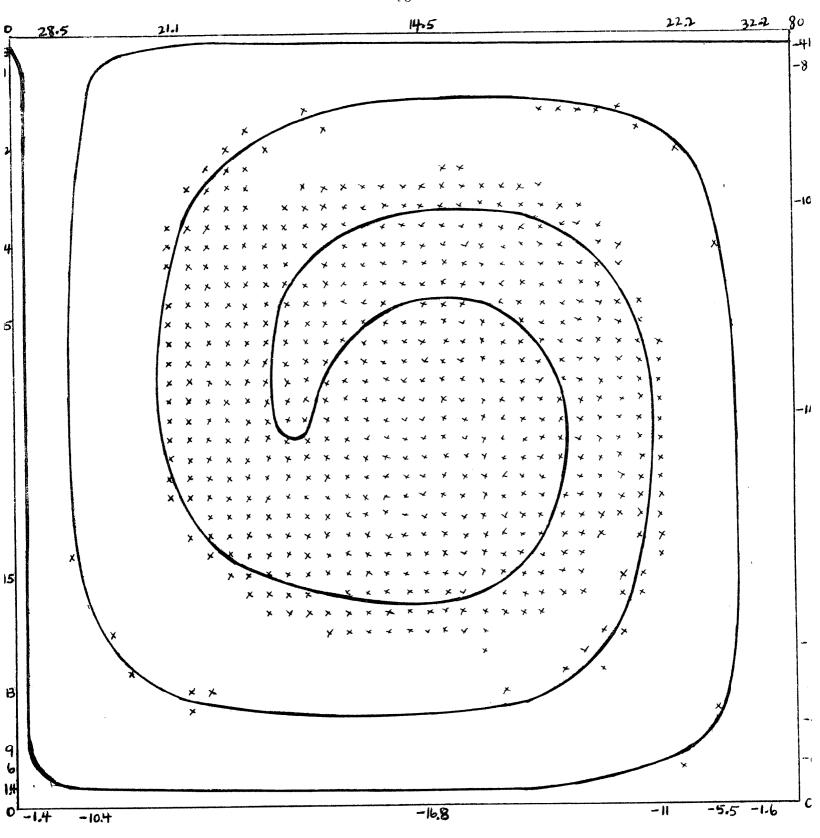


FIGURE 4.6 Equivorticity curve $\omega = 1.630$ for Reynolds number 100000 and h = 1/40. At crossed points vorticity is between 1.6 and 1.7.

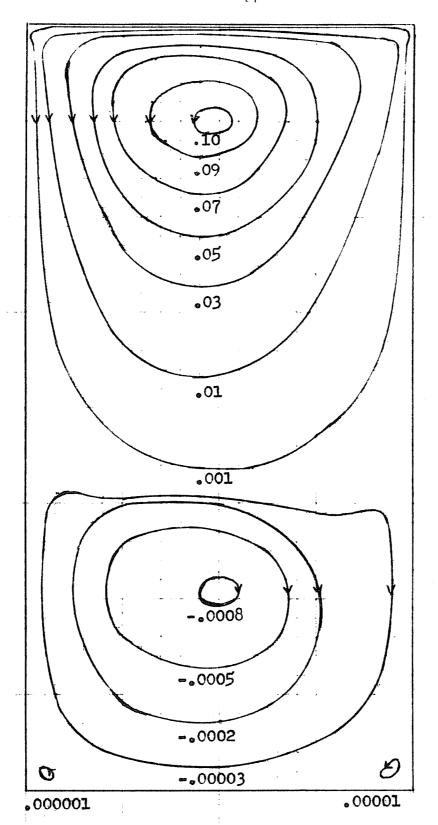


FIGURE 4.7 Streamlines for Reynolds number 10 with h = 1/40 for a 2 by 1 rectangular cavity.

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APPENDIX

CDC 3600 FORTRAN PROGRAM FOR CAVITY FLOW PROBLEMS

D. SCHULTZ

DEFINITIONS OF PROGRAM VARIABLES

OMA = VORTICITY VALUES

PSI = STREAM VALUES

N = NUMBER OF VERTICAL SPACES IN THE GRID

M = NUMBER OF HORIZONTAL SPACES IN THE GRID

R = REYNOLD'S NUMBER

H = GRID SIZE

FPS = TOLERANCE FOR INNER-AND OUTER-ITERATIONS

Cl = WEIGHTING FACTOR FOR OMA

F1 = WEIGHTING FACTOR FOR PSI

RY = RELAXATION FACTOR FOR OMA EQUATIONS

NM = NUMBER OF OUTER-ITERATIONS

NCOURT = NUMBER OF INNER-ITERATIONS

WO, W1, W2, W3, W4 = COEFFICIENTS FOR THE VORTICITY EQUATION

ISTOP = SWITCH TO INDICATE CONVERGENCE

```
THREE TANCETS
        11[7] ATTEM DSI(50.4).).OMA(50.50).SVPAI(50.50).CVOMA(50.50).SVC()[(5
        TOTAL MONITUST & MOMPLUST & MY & MP
        REAL BUOSNOM
411(
        FORMAT(212)
        MDLIJS1=M+1
        MMFSH=M-1
        \cup J \Lambda ! = \cdots
        NP[Hc]=N+I
        NMF CH=N-1
        H=] \cdot /M
        H2=H*H
        5. P.S = . 001
      INITIOLIZE VECTORS
        M7=()
        MP=5
        ISTOP=0
        \Omega = 50
        RW=1.
        Cl=0
        F1 = 0
        CONTINUE
104
        PRINT 2323,C1
2323
        FORMAT(1H1,F8.2)
        DO 1 I=1,50
        no 1 J=1,50
        O = (U, I) THCV2
        SVPSI(I,J)=0
        SVOMA(I,J)=0
        Del(I,J)=0
         0 = (L \cdot I) \Delta MO
        NM=U
        F2 = 1 - F1
        C2 = 1 - C1
      MEGIN LOOP FOR OUTER ITERATIONS
Ć
    SAVE VORTICITY FUNCTION FROM PREVIOUS OUTER ITERATION
23
        DO 40 [=] NPLUS1
        DO 40 J=1, 19LHS1
        (LeI) AMO=(LeI) THOVE
40
        NIM = NIM + ]
        MCOUNT = O
(
    REGIN INNER ITERATION FOR STREAM FUNCTION
        COMPUTE STREAM FUNCTION FOR INNER REGION
        DO 2 1=3.NMESH
        DO 2 J=3,MMESH
        (L, I) I P Q = (L, I) I P Q V P
        PSI(1,J) = (-8*PSI(1,J)) + .45*(PSI(1,J-1) + PSI(1,J+1) + PSI(1+1,J) +
2
      1PSI(I+1,J)+H2*OMA(I,J))
       COMPUTE STREAM FUNCTION ON TOP AND BOTTOM INNER BOUNDARY LINES
        DO 3 I=2.N
        PSI(I,2)=(.25*PSI(I,3))
3
        PSI(I,M) = .25*PSI(I,MMFSH) + .5*H
\subset
        COMPUTE STREAM FUNCTION ON LEFT AND RIGHT INNER BOUNDARY LINES
       DO 4 I=3, MMFSH
```

```
** ( * · · ) = -
                    ( • ) takincl ( a • [ ) )
                   ( • 5 PRD _ I ( N-I • I ) )
        DC1(M.1)=
     THE LEADS IN STING TION FOR CONVERGENC
       TO 5 [=3, MMESH
        DO F J=3, MMESH
       DIFF=ABSE(SVPSI(I.J)-PSI(I.J))
        IF (DIFF .GT. FPS) GO TO 6
     ___COMTINUE_
       RECALCULATE STREAM FUNCTION USING WEIGHTING
       DO 222 I=3 NMESH
       DO 222 J=3, MMESH
       PSI(I,J) = F1 \times SVPSI(I,J) + F2 \times PSI(I,J)
222
       DO 114 I=2, "MESH
      IF(PSI(I,M))28,114,114
1]4
        CONTINUE
       GO TO 200
      NCOUNT=NCOUNT+1
        IF(NCOUNT .GT. 100) GO TO 8
      GO TO 11
     TEST STREAM FUNCTION FOR DIVERGENCE
        IF(DIFF •GT• 10) GO TO 28
       PRINT 93
93
       FORMAT(1H1,11H PSI VALUES)
       CALL PRNTLST(PSI)
10
       FORMAT (10F11.6)
        NCOUNT = 0
      GO TO 11
28
        PRINT 81
       FORMAT(13H PSI DIVERGED)
81
       CALL PRNTLST (PSI)
       CALL PRNTLST (OMA)
       GO TO 699
    BEGIN INNER ITERATION FOR VORTICITY
200
       NCOUNT = 0
30
        HCONST=C2*(-2.7H2)
C
       COMPUTE VORTICITY ON BOUNDARY LINES USING WEIGHTING
       TOP AND BOTTOM BOUNDARY LINES
       DO 12 I=1, NPLUS1
       OMA(1,1)=C1*OMA(I,1)+HCONST*PSI(I,2)
       OMA(I,M+1)=C1*OMA(I,M+1)+HCONS[*(PSI(I,M)-H)]
12
           LAFT AND RIGHT BOUNDARY LINES
C
        00.13 I = 2.4 M
       OM^{(1,1)} = HCONST*PSI(2,I) + C1*OMA(1,I)
        OMA(N+1,I) = HCONST*PSI(N,I) + C1*OMA(N+1,I)
       CONTINUE
\overline{C}
       COMPUTE COEFFICIENTS FOR VORTICITY EQUATIONS
       COMPLETE ONE SWEEP OF INTERIOR
       90.14 I = 2.0
       DO 14 J=2,M
       A1=PSI(I+1,J)-PSI(I-1,J)
        51=PSI(I,J+1)-PSI(I,J-1)
       A=ABSE(A1)
       B=ARSF(R1)
       W0 = 4 + (A + B) * (R/2)
         IF(A1.GF. 0)15,16
```

```
15
                     | W2=1+(ワ/2)*A
                   144 = 1
                   20 70 17
16
                   1 \cdot 7 = 1
                   W4=[+A*(P/2)
1/
                   1F(P1.6F. ")14,19
18
                   보:=1+P*(요/2) __
                   AN TO SO
(...
                   -- 1 - 1 + P # (P / 2 )
                   142 = 1
24
                   ILellaMO=(LellaMOV2
                    18 (18TOP .FO. 1)GO TO 305
               O(1 - 1) = O(1 - 1) 
              1 + (M4/W0) *OMA(I,J-1)) *RW+(1-RW) *OMA(I,J)
                   60 TO 14
C
                   CHECK TO SEE IL BIEFLRENCE EQUATIONS ARE SATISFIED TO •001
305
                   1+(W4/W0)*OMA(I,J-1))-OMA(I,J)
                   DIF=ABSF(DIFF)
                   IF(DIF .GT. EPS1)282,14
                   PRINT 183,1,J
282
                   GO TO 700
14
                   CONTINUE
                   IF (ISTOP •FO• 1) GO TO 700
             TEST VORTICITY FOR CONVERSENCE
                   DO 21 I=2,N
                 DO 21 J=2,M
                   DIFF=AdSF(SVOMA(I,J)+OMA(I,J))
                    IF(DIFF .GE. FPS) GO TO 22
                CONTINUE
21
                   RECALCULATE VORTICITY
                                                                                                USING WEIGHTING
                   DC 144 I=2,N
                   00.144 J=2.9M
                     OMA(I,J)=C1*SVOMA(I,J)+C2*OMA(I,J)
144
                    JM = JM + 1
\subset
                   PRINT OUT EVERY 4 OUTER ITERATES
                    IF (JM • EQ • 4)89 • 59
89
                    JM = 0
                   PRINT 79.NM
79
                   FORMAT(181,12,17% OUTER ITERATIONS)
                   DRINT 91
                   CALL PRNTLST (PSI)
                   PRINT 92
                   CALL PRNTLST (OMA)
              TEST OUTER ITERATIONS FOR CONVERGENCE
\subset
59
                   CONTINUE
                   DO 45 I=1,NPLUS1
                   DO 45 J=1,MPLUS1
                   DIFF=AHSF(SVOUT(I.J)-OMA(I.J))
                   IF(DIFF .GT. FPS) GO TO 7
                    CONTINUE
45
                   M.7 = ()
                   MD = 8
                      PRINT 99,NM
```

```
3367
        THOMATOTEL APPROPRIENT CONVENCED IN . . LA.
        DOINT OF
()]
        FORTHT (IX . I I to DS, V d tiks)
        CALL BRAIL T (DSI)
        HICTMIT 43
92
        OF MAT (1141 - 141) OF GALUES)
        CALL PRHTLST (OMA)
        FPSI=.UO1
        RMAX = 0
        ISTOP=1
(
        CHECK TO SEE IF AIRE-RENCE OUATIONS OR STREAM FUNCTION ARE SATISFIF
\subset
       A TOLERANCE OF .001
        00 181 11=3.NM. Sr
        DO 181 JJ=3, MM+ Sm
        RES=ARSE(PSI(II,JJ)-SVPSI(II,JJ))
        IF(RES .GT. RMAX)301,302
301
        PMAX=RES
302
        CONTINUE
        -LU, II) I29+(LU, I-II) I29+([+LU, II) I29+(LU, I+II) I29+(LU, I+II) I29*4-=A
      (1.1.)
        R=-H*H*OMA(II,JJ)
        D = A - B
        IF(D •GT• FPS1) GO TO 182
181
        CONTINUE
        GO TO 90
182
        PRINT 183, 11, 11
        FORMAT(1H1,47H DIFFERENCE FOU. NOT SETISLIED AT POINT (,12,1H,,12
183
      1,1H))
         GN TN 099
Ċ
      TEST OUTER ITERATIONS FOR DIVERGENCE
        IF(DIFF •GT• 100)199,23
12
        MCOUNT=MCOUNT+1
        IF (MCOUNT .GT. 300) 60 TO 24
        GO TO 90
C
      TEST VORTICITY
                           FOR DIVERGENCE
24
        IF(01+F •GT• 10) GO TO 29
        PRINT 94
        FORMAT (1H1, 14H OMAEGA VALUES)
94
        CALL PRNTLST (OMA)
        PRINT 91
        CALL PRNTLST (PSI)
 32
        FORMAT(10F11.6)
        C=TMINDIM
        GO TO 90
29
        PRINT 82
8.2
         FORMAT(13H OMA DIVERGED)
        CALL PRNTLST (PSI)
        CALL PRNTLST (OMA)
        GO TO 699
199
        DRIMT 189
        FORMAT (26H OUTER ITERATIONS DIVERGED)
180
700
         CONTINUE
        PRINT 303, RMAX
 303
        FORMAT(1H1,17H PSI CONVERGED TO, E12.4)
699
        CONTINUE
```