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# Ensemble Joint Sparse Low Rank Matrix Decomposition for Thermography Diagnosis System

Junaid Ahmed, Bin Gao, Senior Member, IEEE, Wai Lok Woo, Senior Member, IEEE, Yuyu Zhu

Abstract-Composite is widely used in the aircraft industry and it is essential for manufacturers to monitor its health and quality. The most commonly found defects of composite are debonds and delamination. Different inner defects with complex irregular shape is difficult to be diagnosed by using conventional thermal imaging methods. In this paper, an ensemble joint sparse low rank matrix decomposition (EJSLRMD) algorithm is proposed by applying the optical pulse thermography (OPT) diagnosis system. The proposed algorithm jointly models the low rank and sparse pattern by using concatenated feature space. In particular, the weak defects information can be separated from strong noise and the resolution contrast of the defects has significantly been improved. Ensemble iterative sparse modelling are conducted to further enhance the weak information as well as reducing the computational cost. In order to show the robustness and efficacy of the model, experiments are conducted to detect the inner debond on multiple carbon fiber reinforced polymer (CFRP) composites. A comparative analysis is presented with general OPT algorithms. Not withstand above, the proposed model has been evaluated on synthetic data and compared with other low rank and sparse matrix decomposition algorithms.

*Index Terms*— CFRP composites, optical thermography, eigen decomposition, joint low rank sparse decomposition, concatenated matrix factorization, weak signal detection.

### I. INTRODUCTION

THE usage of CFRP in the aerospace and aircraft industry is increasing hugely owing to its unique characteristics as lightweight, stiffness, and resistance to corrosion. For quality assurance to monitor the health and quality of the composite becomes ever more important [1]. The composites are manufactured by sandwiching different layers. For good quality, the layers should have strong bonding. However, due to the manufacturing limitations and installation procedure, defects become inevitable. The most commonly found defects in the composites are debonds and delaminations [2]. These defects occur on the inner part of the composite and are not easy to be detected. Therefore, nondestructive testing (NDT) and structural health monitoring (SHM) is necessary to be conducted.

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Wai Lok Woo is with Department of Computer and Information Sciences Faculty of Engineering and Environment, Northumbria University, Newcastle upon Tyne, U.K. In [3], Poudel *et al.* used the NDT technique for the defect detection and analysis of composite repairs. In [4], Meola *et al.* reviewed the importance of NDT based methods for defect analysis in the composites. The NDT techniques usually use different external sources for defect analysis. Based on this principle, the NDT can be categorized as eddy current based NDT [5], ultrasonic based NDT [6], acoustic emission-based NDT [7], microwave-based NDT [8]. Nowadays, the popular NDT method for composite defect detection is the optical pulse thermography (OPT) [9]–[12]. It is a fast and widearea inspection technique and more detailed review of OPT system can be found in [13], [14].

In [2], Maierhofer *et al.* discussed two modes of OPT i.e. reflection and transmission modes. A more detailed description of the type and usage of the excitation sources for the OPT can be found in [15], [16]. The OPT uses an excitation source to induce temperature variation in the composite. If defects exist, irregular patterns occur and are captured by the infrared camera. These thermal frames in raw form contain a large degree of noise while the defects information is not clear. To improve the contrast of defects and remove noise, the image and video processing algorithms are utilized [17]–[21].

The generally used image pattern analysis technique for defect detection by OPT system is the principal component analysis (PCA) [22]-[24]. It is based on low rank estimation using singular value decomposition (SVD). In [25], independent component analysis (ICA) algorithm is proposed to further enhance the thermal contrast. In [26], thermal signal reconstruction (TSR) algorithm is proposed. It works on polynomial fitting in the logarithmic domain. In [27], [28], pulse phase thermography (PPT) algorithm is proposed for defect detection by analyzing the defects information in the frequency domain. In [29], Yuanlin et al. proposed a novel polynomial fitting coefficient algorithm. It is based on the mixture of fitting time derivative and the coefficient algorithm. In [30], Yousefi et al. proposed a candid covariance-free incremental principal component thermography (CCIPCT) algorithm. The algorithm is an extension to the PCA by decreasing its computational load and increasing the performance. In [31], Lopez et al. evaluated the performance of the TSR algorithm against the partial least square thermography (PLST) technique. The comparison is carried out for CFRP composite debond detection. In [32], Junyan et al. proposed a hybrid algorithm based on the simulation annealing and nelder-mead simplex search. In [33], Zhang et al. proposed an algorithm for feature embedding. The algorithm utilizes the concatenated feature space to perform the low rank sparse matrix approximation. In [34], Ishikawa et al. proposed an extension to the PPT algorithm. They use phase difference between the defect and nondefect regions at the high frequencies for defect quantification. The work [35], [36] proposed a novel sparse principal component thermography (SPCT) algorithm based on PCA [22] for defect detection in CFPR composites using optical thermography. The algorithm in [35] is quite simple and robust for flat shaped CFRP specimens. However, it is not validated for complex and irregular

shape CFRP specimens as well as the varying depths. From the aspect of low rank matrix factorization (LRMF), the algorithm [35] is a twoterm decomposition algorithm. However, the proposed algorithm optimizes the low rank and sparse data jointly in a concatenated feature space in a tri-decomposition framework. The proposed algorithm is tested for different specimens with different shapes as well as varying depth for CFRP specimen. In addition, the proposed algorithm is validated on synthetic data with comparison of other low rank sparse matrix decomposition algorithms. In [37], it presented and compared three different matrix factorization algorithms for defect detection using thermal NDT. The three algorithms include PCA, non-negative matrix factorization and archetypal analysis. All methods are tested on thermographic NDT data and analysis is presented. In [38], authors further test more algorithms on the thermal NDT data. Moreover, wider applications of the thermal NDT are described such as arts, archelogy, and civil structures. The matrix decomposition algorithms are evaluated for these applications and results are analyzed. In [39], Feng et al. proposed a hybrid algorithm based on the TSR and region growing technique for the task of debond detection in the CFRP composites. In [40], Peng et al. proposed a multilayer architecture utilizing the ensemble variation based tensor factorization (EVBTF). The algorithm is tested for debond detection in CFRP composites. In [41], Ahmed et al. proposed a sparse-mixture-of-gaussian (S-MOG) algorithm for debond detection in CFRP composites. The algorithm utilizes the multilayer structure to mine the features for thermographic image enhancement.

The proposed algorithm falls into the category of tri-decomposition based algorithms. In [42], Zhou et al. proposed a three-term decomposition model called stable principal component pursuit. In this model, the noise term is modeled to be independent identically distributed. The model is solved iteratively by solving the sparse term with a difference equation and the low rank term is estimated by using the least square method. In [43], Aravkin et al. proposed variation of the stable principal component pursuit method. In this model, it decomposes the matrix into the two parts as they are solved sequentially by projected and accelerated gradient methods. In [44], Oreifej et al. proposed a novel model for the background and foreground segmentation problem in video sequences. They solve the three-term decomposition model in an iterative manner in the framework of the augmented Lagrangian multiplier method. In [45], Zhang et al. proposed a tri-decomposition model in the framework of low-rank matrix recovery and completion. It decomposes the observed data into the clean data, sparse data and noise data. It is tested in a variety of face images and surveillance videos in the framework of image denoising. These algorithms utilize a single feature space for the optimization of the tri-decomposition model containing the observed raw data using the augmented Lagrangian multiplier method. The proposed method utilizes the concatenated feature space for the low rank matrix decomposition using the residual and sparse data along with the observed raw data. The low rank information from the concatenated feature space is able to extract the weak target defect information as the defects information lies in the low rank as well as sparse space. In addition, the proposed method solves the tridecomposition model by developing an expectation-maximization (EM) framework for the ensemble joint sparse low rank matrix decomposition (EJSLRMD).

As the defects depth increases, the detection performance decay. For the composite specimen with an irregular shape, the general OPT algorithms give poor performance [41]. The algorithm of [41] has good reasonable results whereas its computational cost is quite high due to the multilayer sparse modelling structure. To alleviate this problem, we propose EJSLRMD algorithm. The proposed algorithm models the low rank and sparse data jointly in a concatenated feature space. Since the defects information mostly presents in the sparse and low rank space, it is possible to mine the low rank feature in a concatenated feature space with the raw data before sparse modeling. To reduce the computational cost we chose the most significant eigen-features for the sparse modeling. The proposed algorithm is able to detect weaker and deeper defects. In order to show its efficacy, the algorithm is conducted for debond defects detection in a different structure of CFRP composites. The visual analysis along with F-score [40] comparison are presented with generally used OPTNDT algorithms. In addition, the proposed algorithm is validated on the synthetic data with different noise configurations.

The rest of this paper has been organized as follows: The proposed algorithm is described in Section 2. The experimental setup and information about the CFRP specimen are given in Section 3. Results and discussions are elaborated in Section 4. Finally, conclusions are drawn in Section 5.

#### II. THE PROPOSED METHODOLOGY

# A. Proposed Algorithm

Given the data tensor containing the thermographic sequences  $D \in \mathbb{R}^{m \times n \times k}$  where (m, n) denote the spatial resolution of the frame and k represents the number of the frame. Firstly, we convert it into a matrix form by representing each (m, n) spatial frame as a vector for i frames. Secondly, this matrix can be modeled into a multilayer structure [40], [41] of low rank matrix L, sparse matrix S and noise matrix E as:

$$D^1 = L^1 + S^1 + E^1 \tag{1}$$

For the second layer decomposition, it can be expressed as:

I

 $D^{2} =$ 

$$f^1(D^1) + L^2 + S^2 + E^2 \tag{2}$$

In general, for the  $i^{th}$  layer, the deep decomposition can be written as:

$$D^{i} = f^{i-1}(D^{i-1}) + L^{i} + S^{i} + E^{i}$$
(3)

where  $f^i(D^i)$  is the activation used in the multilayer low rank sparse data modelling. This structure is portrayed in Fig. 1.

Fig. 1 shows the overall schematic block diagram of the proposed model. It is divided into four core parts for better interpretation. The orange blocks represent the input thermal sequences. The blue blocks represent the concatenated feature space eigen decomposition. The green blocks represent the model for the probabilistic robust matrix factorization algorithm. Finally, the red block is the output. Given the input data and initializations of the sparse matrices, the concatenated eigen decomposition is performed as shown in the blue blocks on the top of Fig. 1. In the next step, the sparse matrix decomposition is performed and its probabilistic model is shown by the green block in Fig. 1. This process of ensemble joint sparse low rank matrix decomposition is solved in an iterative manner where the concatenated low rank component is solved by eigen decomposition and sparse component is solved by expectation maximization approach as shown in the middle blocks of the Fig. 1. Finally, the overall process is represented as a multilayer ensemble architecture of the low rank and sparse factorization as shown in the bottom blocks of Fig. 1. The whole structure is applied to extract the weak defect information on CFRP composites using the optical thermography.

The previous study does not involve or leverage the sparse factors for the spatial resolution of the thermal data. Sparseness refers to a representational scheme where only a few units (out of a large population) are effectively used to represent typical data vectors. In effect, this implies most units taking values close to zero while only few take significantly non-zero values. The sparse factors enforce the solution to consider only the significant region where the defect may lie within the surrounding background. For data with sparse outliers are partially contaminated by noise of overwhelming magnitude, sheer low-rank assumption cannot fully capture its complex structure. Therefore, (1) can be considered as combination of sparse patterns (e.g. hot spots) and non-sparse patterns. Thus, to extract the defect information from the thermographic data, we propose the following optimization problem [44], [45]:

$$\min_{L_{s}} \left\{ \left\| L^{i} \right\|_{*} + \Lambda \left\| S^{i} \right\|_{2} + \left\| D^{i} - L^{i} - S^{i} \right\|_{F}^{2} \right\}$$
(4)

where  $\Lambda$  is the regularizing parameters for S,  $\|.\|_2$  represents the  $l_2$ norm,  $\|.\|_*$  represents the nuclear norm for low rank term L, and  $\|.\|_F$ represents the Frobenius norm. Using the regularizing framework, we relax the above problem using convex proxies. In addition, for any non-singular matrix,  $S = AS^{-1}SB^{T}$  holds. The problem (4) can be

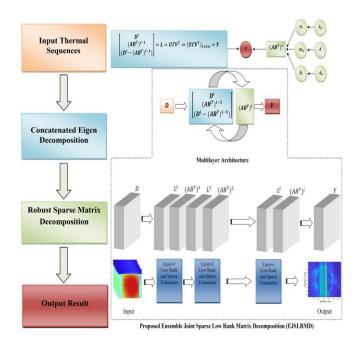


Fig. 1. The proposed model description

reformulated as:

 $\min_{LAB} \left\{ \left\| L^{i} \right\|_{*} + \Lambda_{a} \left\| A^{i} \right\|_{2}^{2} + \Lambda_{b} \left\| B^{i} \right\|_{2}^{2} + \left\| D^{i} - L^{i} - (AB^{T})^{i} \right\|_{F}^{2} \right\}$ (5)

where  $\Lambda_a, \Lambda_b$  are the regularizing parameters for A, B. The problem of (5) is solved in two steps. In the first step, we solve for the L which is the low rank term. In the second step, we solve for  $S = AB^{i}$  which represents the sparse term. The steps are elaborated in graphical form as shown in Fig.1. For the low rank term, given the data matrix D and initial matrices of A, B, we propose a concatenated eigen decomposition for the low rank term:

$$L^{i} = \begin{bmatrix} D^{i} \\ D^{i} - (AB^{T})^{i-1} \\ (AB^{T})^{i-1} \end{bmatrix}$$
(6)

where *i* represents the layer number. For the problem of defect detection in the CFRP composite structure by using optical thermography, the thermal video sequences contain multiple frames of the same specimen on different transient responses. Based on the analysis in [41] and [40], the defect information is mostly present in the sparse and low rank components of the decomposition. By concatenating the original data with residual and sparse data for the eigen decomposition, it is able to extract more information of the

defects as compared to the simple eigen decomposition without concatenation which can be seen in the results of PCA [22] Fig. 4. In particular, this data goes into the sparse decomposition algorithm of [46] which further removes the noise and modifies the sparse data in an iterative manner. By using the concatenated feature space in a joint sparse and low rank decomposition, it significantly enhances the extraction of weak defect information.

By concatenating the sparse data, two benefits can be achieved. Firstly, we keep intact the original raw features in the low rank estimation. This enforces that the estimated low rank features do not significantly deviate from the original features. Secondly, we use the sparse data and residual data for low rank estimation. It significantly embeds the sparse information into the low rank space which subsequently allows the algorithm to extract the target weak defect information from both low rank space and sparse space in a joint optimization framework by using the concatenated feature space. We solve the problem of (6) by using eigen decomposition technique as: Ιİ

$$= U\Gamma V^T \tag{7}$$

where U, V are the left and right eigenmatrices and  $\Gamma$  is the diagonal matrix containing the eigen values. The first six principal eigenvectors are chosen to represent the low rank term. This setting is based on repeated experimental analysis and it is observed that six eigenvectors can already contain the most useful low rank information, namely:

 $Y^i = (U\Gamma V^T)_{1 to 6}$ (8)For  $S = (AB^T)$ , we solve the following optimization problem [46]:  $(AB^T)^i = \arg \min_{A,B} \left\{ \|Y^i - (AB^T)^{i-1}\|_F^2 + \Lambda_a \|A^{i-1}\|_2^2 + \Lambda_b \|B^{i-1}\|_2^2 \right\}$  (9) It should be noted that the most expensive step is sparse modelling. As only six principal eigenvectors are used to represent the low rank term,

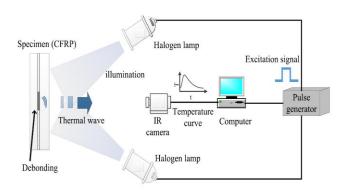


Fig. 2. Block diagram of the OPT system

the computational cost will be significantly reduced. We solve the problem of (9) for each layer *i* by using the probabilistic robust matrix factorization (PRMF) algorithm of [46]. The algorithm of [46] utilizes the conditional expectation minimization (CEM) algorithm of [47] to update the A, B in an iterative manner. First, we decompose Y containing the concatenation information as following matrix factorization problem [46]:

$$Y = AB^T + E \tag{10}$$

$$\begin{array}{l}a_{ij}|\Lambda_a \sim \aleph(a_{ij}|0,\Lambda_a^{-1}) \tag{11}\\b_{ij}|\Lambda_b \sim \aleph(b_{ij}|0,\Lambda_b^{-1}) \tag{12}\end{array}$$

$$b_{ij}|\Lambda_b \sim \aleph(b_{ij}|0,\Lambda_b^{-1}) \tag{12}$$

where E is the noise matrix and  $a_i$  be the  $i^{th}$  row of A and  $b_i$  be the *j*<sup>th</sup> row of *B*. Assuming noise follows the Laplacian distribution. This implicates:

$$p(E|\Lambda) = \left(\frac{\Lambda}{2}\right)^{mn} \exp\{-\Lambda \|E\|_1\}$$
(13)

Let A, B be the parameters to be estimated. A,  $\Lambda_a$  and  $\Lambda_b$  are the hyper parameters. The MAP theory and Bayes theorem:

$$p \propto p(Y|A, B, \Lambda)p(A|\Lambda_a)p(B|\Lambda_b)$$
(14)

where

$$logp(A, B|Y, \Lambda, \Lambda_a, \Lambda_b) = -\Lambda ||Y - AB^T||_1 - \Lambda_a ||A||_2^2 - \Lambda_b ||B||_2^2 + C (15)$$

where C is the constant term. The problem of (15) is the same as minimizing the following problem:

$$\min_{A,D} \|Y - AB^T\|_1 + \Lambda_a \|A\|_2^2 + \Lambda_b \|B\|_2^2 \tag{16}$$

To solve this problem, a leveled hierarchical form of a Laplacian distribution is used. Let y be the Laplacian random variable, its probability density function (pdf) can be given as:

$$p(y|a, l^2) = \frac{l^2}{2} \exp(-l^2|y-a|)$$
(17)

The Laplacian distribution can be represented as a mixture-of-gaussians as:

$$L(y|a, l^2) = \int_0^\infty \aleph(y|a, m) \, Expon(m, l^2) dm \tag{18}$$

where  $Expon(m, l^2)$  is the exponential distribution term. To accommodate this, a matrix  $M = [m_{ij}] \in \mathbb{R}^{m \times n}$  is used whose each element follows exponential prior. This variable relates the  $l_1$  term to the  $l_2$  term and hence we can have a closed form solution.

Let  $a_i$  be the  $i^{th}$  row of A and  $b_j$  be the  $j^{th}$  row of B. The matrix factorization can be formulated as:

$$y_{ij}|A, B, M \sim \aleph \left( y_{ij} | a_i^T b_j, m_{ij} \right)$$
(19)

$$a_{ij}|\Lambda_a \sim \aleph(a_{ij}|0,\Lambda_a^{-1}) \tag{20}$$

$$b_{ij}|\Lambda_h \sim \aleph(b_{ij}|0,\Lambda_h^{-1}) \tag{21}$$

$$m_{ij} | \Lambda \sim Expon(m_{ij} | \Lambda/2)$$
(22)

To estimate A, B, conditional EM algorithm is used [47]. The EM algorithm iterates between two steps, E-step and M-step. For the E-step, the Q-function is solved. Given the initial estimates be  $\hat{\theta} = [\hat{A}, \hat{B}]$ , namely

$$Q(B|\hat{\theta}) = E_M[logp(B|\hat{A}, Y, M)|Y, \hat{\theta}]$$
(23)

Taking log on both sides and ignore the terms which do not relate to Q.

$$\log p(Y|B, A, M) + \log p(B) = -\frac{1}{2} \sum_{i}^{m} \sum_{j}^{n} \{m_{ij}^{-1}(y - \hat{a}_{i}^{T}b_{j})^{2}\} - \Lambda_{b} \sum_{j}^{n} b_{j}^{T}b_{j} + C$$
(24)

It can be seen that  $m_{ii}^{-1}$  obeys an inverse Gamma distribution.

$$E\left[m_{ij}^{-1}|Y,\hat{A},\hat{B}\right] = \frac{\sqrt{A}}{|u_{ij}|} \triangleq \langle m_{ij}^{-1} \rangle$$
(25)

where  $u_{ij} = y_{ij} - (ab^T)_{ij}$ . Next, in the M-step, the parameter *B* is updated. This is done by maximizing the Q-function. To achieve this take the partial derivative of Q-function with respect to  $b_j$  and set it to zero. The update rule can be set as:

$$b_j = \left(\hat{A}^T \Omega_j \hat{A} + \Lambda_b I_u\right)^{-1} \hat{A}^T \Omega_j y_{.j}$$
(26)

where  $\Omega_j = diag(\langle m_{1j}^{-1} \rangle, \dots, \langle m_{mj}^{-1} \rangle)$  and y is the  $j_{th}$  coloumn of Y. Following the same convention, the update formula for a can be found as:

$$a_i = (B^T \Lambda_i \hat{B} + \Lambda_a I_u)^{-1} \hat{B}^T \Lambda_i y_i.$$
<sup>(27)</sup>

where  $\Lambda_i = diag(\langle m_{i1}^{-1} \rangle, \dots, \langle m_{in}^{-1} \rangle)$  and  $y_i$ . is the  $i_{th}$  row of Y. As the data Y consists of only six principle eigenvectors, the CEM algorithm based on experimental analysis updates A, B in only two iterations.

The stopping condition for the proposed EJSLRMD problem is set as:

$$\sum_{i} \frac{(u_{ij}^{i} - u_{ij}^{i-1})}{u_{ij}^{i-1}} < \in$$
(28)

The term  $\in$  represents the tolerance level which has been selected to be  $10^{-6}$  based on the independent Monte-Carlo test. The complete step-by-step description is tabulated in Table. 1.

# TABLE I THE PROPOSED ENSEMBLE JOINT SPARSE LOW RANK MATRIX DECOMPOSITION (EJSLRMD)

- 1. Input Data  $D \in \mathbb{R}^{m \times n \times k}$
- 2. Convert the tensor D into matrix form.
- 3. Initialize the parameters  $\Lambda_a$ ,  $\Lambda_b$  as 1 and A, B randomly.
- 4. For each layer do;
- 5. Solve for L using the (6) to (7).
- 6. Solve for A and B using CEM algorithm.
- 7. E-Step: for A and B  $\langle m_{ij}^{-1} \rangle = \frac{\sqrt{A}}{|u_{ij}|}$
- 8. M-Step:
- 9.  $\boldsymbol{b}_{i} = \left(\widehat{A}^{T} \boldsymbol{\Omega}_{i} \widehat{A} + \Lambda_{b} \boldsymbol{I}_{u}\right)^{-1} \widehat{A}^{T} \boldsymbol{\Omega}_{i} \boldsymbol{y}_{\cdot i}$
- 10.  $\boldsymbol{a}_i = \left(\boldsymbol{B}^T \boldsymbol{\Lambda}_i \boldsymbol{\widehat{B}} + \boldsymbol{\Lambda}_a \boldsymbol{I}_u\right)^{-1} \boldsymbol{\widehat{B}}^T \boldsymbol{\Lambda}_i \boldsymbol{y}_i.$
- 11. Check the stopping criteria using (28) or go to step 5.
- 12. End for
- 13. Output *L*, *S*

The Matlab demo code can be linked: http://faculty.uestc.edu.cn/gaobin/zh\_CN/lwcg/153392/list/index.ht m

# III. EXPERIMENTAL SETUP

## A. Experiment set-up and specimen details

In an experimental evaluation, Fig. 3 shows the OPT system with the reflection mode configuration [48]. Halogen lamps are used as the source of excitation with the power of 2kW. At the back hand, optical excitation source of ITECH-IT6726G is used which is a ZY - B type source. It comes with adjustable DC power mechanism which can go up to 3kW. The distance between the specimen under test and excitation source is set around 80cm. The A655sc infrared camera is used to capture the time series temperature variations of the specimen. The resolution of the camera is  $640 \times 480$ . The thermal sensitivity of



Fig. 3. The optical pulse thermography system

the camera is  $0.05^{\circ}C$ . In our experiments, we have utilized the sampling frequency of 50Hz.

OPT technology utilizes an external heating source and an infrared camera. The specimen is excited using external sources and the temperature variations are captured. These temperature variations are represented as the time series of the thermographic images. The pulse generator is used to control the frequency of excitation and a computer is applied to store the results. The configuration of the reflection mode is used with the halogen lamps as the source of heating. The halogen lamps and the infrared camera are placed facing the same direction of the specimen as the reflection mode as shown in the schematic block diagram of OPT in Fig. 2.

Five different CFPR composite specimen are prepared for the experimental validation of the proposed algorithm. The CFRP composites were acquired from the Chengdu Aircraft Design Institute which is a part of the China Aviation Industry. These specimen were used in the design and manufacturing of the aircraft components. The

first two specimens are flat surface with a rectangular shape. The remaining three samples have the V shape irregular surface. All the specimen have debond defects of different diameters and depths. The more detailed information about the specimen and defects can be found in Table. 2.

TABLE II INFORMATION ABOUT THE CFRP SPECIMEN

Number	Defect Profile	Dimension(mm)	Defect Information(mm) Top Depth, Bottom Diameters	Picture
1		250×250×24.2	1, 2 2,4,6,8,10,12,16,20	
2		250×250×22.2	2, 2.5 2,4,6,8	4#
3	S0m S0m S0m S0m S0m S0m S0m S0m	100×100×80	2, 2.25, 2.5,2.75 2, 3	
4	None Some Some Some Some Some Some Some Som	100×100×80	0.5,0.75,1,1.25,1.5,1.75 2, 3	
5	Som Som Som Som Som Som Som Som Som Som	100×100×80	1.5,1.75,2,2.25,2.5,2.75 9, 10	

# IV. EXPERIMENTS ANALYSIS

The visual results along with the quantitative results are presented. The comparative analysis is carried out with the general OPT algorithms to show the efficacy and efficiency of the proposed algorithm. The quantitative comparison parameters used are F-score and the running (computation) time. The general OPT based NDT algorithms under comparison are PCA [22], PPT [27], TSR [26], EVBTF [40] and S-MoG [41]. All the experiments are carried out in a corei7 computer with a Windows-10 operating system having 8GB RAM. MATLAB2017b software is utilized for all the algorithms evaluation. The comparative results for all specimen are summarized in Table. 3.

The visual comparative results are shown in Fig. 4 in a tabular form. Row 1 shows the comparison results for specimen 1. It is a flat surface rectangular shape specimen. The defect depths are 1mm and 2mm. For this specimen, almost all the algorithms perform well. However, from Fig. 4 (row 1) left to right, it can be seen that strong noise is still present and all algorithms fail to detect the defect with the smallest diameter defects on the right end corner. Nonetheless, the proposed algorithm gives better contrast and resolution result. It detects all the debond defects present on the specimen. Fig. 4 (row 2) shows the results of the second sample with a flat surface and rectangular shape. The defect depths are 2mm and 2.5mm. In comparison, the proposed algorithm gives better contrast and resolution and quantifies more defects than the other algorithms.

Fig.4 (row 3) shows the comparative results for the specimen 3. It is a V shaped irregular surface specimen. The defect depths are (2,2.25,2.5,2.75)mm. From Fig. 4 (row 3) left to right, most algorithms fail in detecting the debond defects. The proposed algorithm is able to give reasonable contrast and resolution results. The proposed algorithm detects all the defects present in the specimen. Fig. 4 (row 4) shows the visual results for CFRP specimen 4. Here, the number of defects are 6. The depths are (0.5,0.75,1,1.25,1.5,1.75)mm. Because of the irregular shape and surface, the performance of these algorithms is quite poor. The proposed algorithm gives better resolution with good contrast results. All the debond defects are successfully detected.

Fig. 4 (row 5) shows the visual results for specimen 5. The number of defects here are 5. The depth of the defects are (1.5,1.75,2,2.25,2.5)*mm*. The diameter of the defects are 9*mm* and 10*mm*. In the comparative analysis, the proposed algorithm detects all the debond defects present on the specimen and shows good resolution and contrast.

The quantitative comparison based on F-score and computation time are tabulated in Table. 3. The last row shows the average percent F-score for all the algorithms along with the average computation time in seconds. On average, the PPT algorithm has the detection efficiency of 63% with 208 seconds in average running time. The average detection rate in terms of percent F-score for the TSR algorithm is 76% with the average time consumption of 494 seconds. The PCA algorithm has the fastest running time of 56 seconds with a reasonable detection rate of 76%. The algorithm of EVBTF gives the highest running time of 970 seconds with a poor detection capability of 40%. The S-MoG algorithm takes an average time of 190 seconds to produce the results with the percent efficiency of 71%. The proposed algorithm gives on average the highest detection rate of 99%. The proposed algorithm takes around on average 76 seconds to be the second-fastest algorithm to PCA. By jointly optimizing the low rank and sparse data in a concatenated manner, it can remove the noise, improve the resolution and increase the detection efficiency.

	PPT[27]	TSR[26]	PCA[22]	EVBTF[40]	S-MoG[41]	EJSLRMD
1						
2						
3	100 200 400 100 200 300 400 500 600	100 200 300 400 100 200 300 400 500 600			100 200 400 100 200 300 400 500 600	

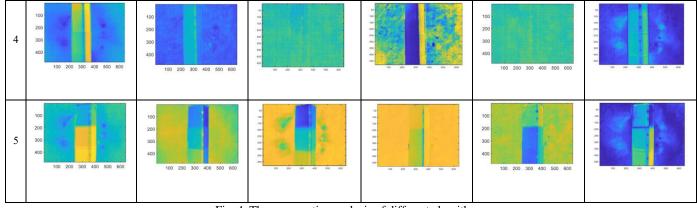


Fig. 4. The comparative analysis of different algorithms. TABLE III

COMPARATIVE RESULTS F-SCORE (LEFT) AND TIME TAKEN (RIGHT IN SECONDS).

Specimen Number	PPT	[27]	TSR	[26]	PCA	[22]	EVB	ΓF[40]	S-Mo	G[41]	EJSLR	MD
1	0.94	135	0.94	271	0.94	43	0.94	1342	0.94	173	1	51
2	0.66	564	0.66	642	0.93	153	0.30	1019	0.93	466	0.93	52
3	0.4	129	0.66	241	0.66	15	0.00	766	0.4	86	1	90
4	0.4	124	0.66	631	0.4	30	0.00	1039	0.4	120	1	93
5	0.75	146	0.88	601	0.88	47	0.75	753	0.88	125	1	95
Average	63%	208	76%	494	76%	56	40%	970	71%	190	99%	76

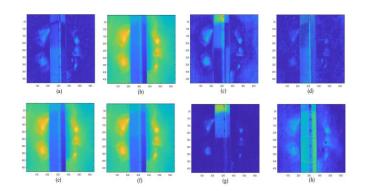


Fig. 5. Comparative results for specimen 4 on different algorithms and their computation time in seconds (a) [52] (156sec) (b) [50] (1986sec) (c) [53] (340sec) (d) [55] (420sec) (e) [42] (464sec) (f) [35] (29sec) (g) [38] (14sec) (h) proposed (93sec)

The proposed model uses the PRMF [46] algorithm for the sparse decomposition step. However, there are other similar algorithms in the literature. In [49], Xiang et al. proposed a matrix factorization algorithm called direct robust matrix factorization algorithm (DRMF). The block coordinate descent approach is proposed to solve the lowrank decomposition problem which is a variation of the singular value decomposition (SVD) and efficient thresholding. In [50], Wang et al. proposed a Bayesian extension to the PRMF [46] model for the image and video processing applications. In [51], Zhao et al. proposed a model for the low rank matrix factorization (LRMF) problem which utilizes the inference based variational Bayes framework. It has been found that these class of algorithms have high computational cost for the problem of defect detection in CFRP composites. In [52], Meng et al. proposed a novel model for the LRMF problem, where they assume the noise to have an unknown probabilistic distribution and estimate it by using mixture of Gaussian (MoG) model. In [53], Cao et al. improved the model of [52] by assuming the noise has mixture of exponential power (MoEP) distribution and propose an expectation maximization algorithm to solve the problem. In [54], Kim et al.

proposed a novel algorithm for the LRMF problem which utilizes the orthogonal matrix decomposition algorithm in the augmented Lagrangian framework. In [55], Lin *et al.* proposed a majorization minimization approach for the problem of LRMF. A surrogate function is used to replace the original problem and the algorithm of linearized alternating direction method with parallel splitting and adaptive penalty (LADMPSAP) is used for its solution owing to its low computation cost.

The algorithm of [49] is a simple and easy way to implement whereas its performance is normal. The algorithms of [50], [51] are based on the Bayesian framework. The class of variational Bayes framework based algorithms for the problem of defect detection in CFRP composites using optical thermography have been analyzed by [40]. These algorithms have poor performance and high computation cost for irregular shape CFRP specimen. The algorithms of [52]-[55] are quite robust and assume the noise has a more complex distribution rather than the Gaussian distribution. These class of algorithms were analyzed in [41]. It has been found that these algorithms for the defect detection problem with the irregular shape specimen fail to perform well. In addition, these algorithms are quite complex and lots of parameters need to be tuned for the solution of a particular problem. Based on this analysis, the PRMF [46] algorithm was selected owing to its simple implementation, less parameter tuning and robustness to fit in the framework of EJSLRMD. In the multilayer architecture of EJSLRMD, it requires more parameters and complex architecture which increases the computational cost as referred in [40] and [41]. The PRMF algorithm in the proposed framework converges significantly fast and simultaneously it is able to recover the signal more accurately with complex noise distributions.

Fig. 5 shows the comparative results on specimen 4 with irregular shape who has six defects on varying depths. Fig. 5 (a) shows the results of the matrix factorization algorithm of [52]. It can be observed from the results that it is difficult to distinguish the defects and background. The computational cost is 156 seconds. Fig. 5 (b) is the result of Bayesian robust matrix factorization algorithm of [50]. The results are over smooth and defects are not clearly visible. The computational cost is significant high of 1986 seconds. Fig. 5 (c)

shows the result of exponential power distribution based algorithm [53]. Here the noise is assumed to have a more complex distribution. However, the algorithm is unable to detect the defects clearly. The computational cost of this algorithm is 340 seconds. Fig. 5 (d) shows the result of matrix factorization algorithm in [55]. However, it is unable to detect the defects and its computational time is 420 seconds. Fig. 5 (e) shows the result of tri-decomposition model of [42] called the stable principal component pursuit. The computational cost is 464 seconds. The result is over smooth and defects are hidden in the background and blurry. Fig. 5 (f) shows the results of state-of-the-art algorithm of [35] called the sparse principal component thermography. The computational time is very less 29 sec. However, as the CFRP specimen has an irregular shape and varying depth the algorithm is unable to detect the defect more clearly. Fig. 5 (g) shows the results of

non-negative matrix factorization algorithm in [38]. This algorithm has least computational cost of 14sec. The algorithm is able to detect at most 3 defects out of 6 with a strong noise present. The last figure shows the result of the proposed algorithm. The computational time is 93 seconds. It can be seen that the proposed algorithm is able to detect all defects clearly with good resolution and reasonable computational cost. For the case of debond detection in CFRP composites with irregular shape and varying depth, the proposed ensemble joint sparse low rank matrix decomposition algorithm provides better quality and detection results under comparison with recent matrix factorization and other infrared non-destructive testing (IRNDT) state-of-the-art algorithms.

TABLE IV

	EXPERIMENTAL	L ANALYSIS	S ON SYNTHE	TIC DATA WI	TH DIFFEREN'	T NOISE CC	NFIGURAT	TONS	
(10)	DCA	[22] DD(		7A[57] VDD	DCA[59] DD1	ME[46] M	G[50] S ]	$M_{0}C[41]$	)

	PCA[22]	RPCA[56]	BRPCA[57]	VBRPCA[58]	PRMF[46]	MoG[59]	S-MoG[41]	Proposed
RRE	1.80e-15	1.76e-8	0.196	1.18e-3	1.56e-5	1.52e-4	2.33e-6	1.98e-8
Time(s)	0.0019	0.0961	46.61	0.0190	0.342	0.160	0.280	0.190
RRE	0.789	3.39e-3	7.99e-2	0.863	7.11e-5	8.44e-5	6.60e-6	7.48e-7
Time(s)	0.0041	0.187	40.11	0.116	0.710	0.310	0.417	0.380
RRE	3.10e-2	5.10e-2	3.19e-2	4.96e-2	3.91e-2	3.14e-2	1.16e-3	6.48e-4
Time(s)	0.0037	0.179	88.69	0.120	0.640	0.310	0.569	0.480
RRE(s)	1.07	0.109	7.66e-2	1	7.44e-2	2.64e-2	4.56e-3	9.04e-3
Time	0.0036	0.180	28.66	0.862	0.622	1.36	1.98	1.56
	Time(s) RRE Time(s) RRE Time(s) RRE(s)	RRE         1.80e-15           Time(s)         0.0019           RRE         0.789           Time(s)         0.0041           RRE         3.10e-2           Time(s)         0.0037           RRE(s)         1.07	RRE         1.80e-15         1.76e-8           Time(s)         0.0019         0.0961           RRE         0.789         3.39e-3           Time(s)         0.0041         0.187           RRE         3.10e-2         5.10e-2           Time(s)         0.0037         0.179           RRE(s)         1.07         0.109	RRE         1.80e-15         1.76e-8         0.196           Time(s)         0.0019         0.0961         46.61           RRE         0.789         3.39e-3         7.99e-2           Time(s)         0.0041         0.187         40.11           RRE         3.10e-2         5.10e-2         3.19e-2           Time(s)         0.0037         0.179         88.69           RRE(s)         1.07         0.109         7.66e-2	RRE         1.80e-15         1.76e-8         0.196         1.18e-3           Time(s)         0.0019         0.0961         46.61         0.0190           RRE         0.789         3.39e-3         7.99e-2         0.863           Time(s)         0.0041         0.187         40.11         0.116           RRE         3.10e-2         5.10e-2         3.19e-2         4.96e-2           Time(s)         0.0037         0.179         88.69         0.120           RRE(s)         1.07         0.109         7.66e-2         1	RRE         1.80e-15         1.76e-8         0.196         1.18e-3         1.56e-5           Time(s)         0.0019         0.0961         46.61         0.0190         0.342           RRE         0.789         3.39e-3         7.99e-2         0.863         7.11e-5           Time(s)         0.0041         0.187         40.11         0.116         0.710           RRE         3.10e-2         5.10e-2         3.19e-2         4.96e-2         3.91e-2           Time(s)         0.0037         0.179         88.69         0.120         0.640           RRE(s)         1.07         0.109         7.66e-2         1         7.44e-2	RRE         1.80e-15         1.76e-8         0.196         1.18e-3         1.56e-5         1.52e-4           Time(s)         0.0019         0.0961         46.61         0.0190         0.342         0.160           RRE         0.789         3.39e-3         7.99e-2         0.863         7.11e-5         8.44e-5           Time(s)         0.0041         0.187         40.11         0.116         0.710         0.310           RRE         3.10e-2         5.10e-2         3.19e-2         4.96e-2         3.91e-2         3.14e-2           Time(s)         0.0037         0.179         88.69         0.120         0.640         0.310           RRE(s)         1.07         0.109         7.66e-2         1         7.44e-2         2.64e-2	RRE         1.80e-15         1.76e-8         0.196         1.18e-3         1.56e-5         1.52e-4         2.33e-6           Time(s)         0.0019         0.0961         46.61         0.0190         0.342         0.160         0.280           RRE         0.789         3.39e-3         7.99e-2         0.863         7.11e-5         8.44e-5         6.60e-6           Time(s)         0.0041         0.187         40.11         0.116         0.710         0.310         0.417           RRE         3.10e-2         5.10e-2         3.19e-2         4.96e-2         3.91e-2         3.14e-2         1.16e-3           Time(s)         0.0037         0.179         88.69         0.120         0.640         0.310         0.569           RRE(s)         1.07         0.109         7.66e-2         1         7.44e-2         2.64e-2         4.56e-3

The proposed algorithm is tested on the synthetic data for modeling different types of noise and results as presented in Table. 4. A series of matrix decomposition based algorithms are compared. The results are quoted in terms of the relative reconstruction error (RRE) and time in seconds. Table. 4. shows that the proposed algorithm is able to recover the mixture of noise more accurately as compared with the other algorithms of PCA[22], robust principal component analysis (RPCA)[56], Bayesian robust principal component analysis (BRPCA)[57], variational Bayesian principal component analysis (VBRPCA)[57], variational Bayesian principal component analysis (VBRPCA)[58], PRMF [46], mixture of Gaussian (MoG) [59], and S-MoG[41]. The best results are highlighted in bold. It can be seen that the proposed algorithm is able to recover the signal with least error when the noise is considered as the complex noise also the time taken is reasonable as compared with other algorithms.

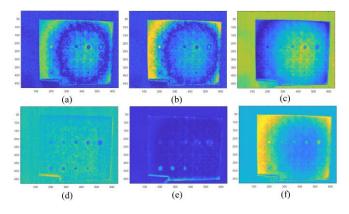


Fig. 6. The inherent layering results for the specimen 1 (a) Layer-1 (b) Layer-2 (c) Layer-3 (d) Layer-4 (e) Layer-5 (f) Layer-6

Fig .6 shows the inherent layering results for specimen 1. The proposed algorithm is able to detect and quantify the defects up to layer 4 for

this specimen. Further layering induces overfitting of the data and the results get worse as can be seen from Fig. 6. (e) and (f).

# V. CONCLUSION

In this paper, a joint low rank sparse modelling algorithm is proposed. The algorithm is evaluated for inner debond defects as well as on synthetic data for modelling the complex noise. By optimizing the low rank and sparse data using the concatenated feature space helps boost the computation speed, estimate the complex noise and detect weaker information defects hidden in background. The quantitative results based on F-score and RRE prove that proposed model performs well in modelling complex noise and quantifying weaker debond defects who presented on the irregular shape CFRP composites. The comparative analysis with general OPTNDT and low rank sparse modelling algorithms proves the efficacy of the proposed model.

In future works, the proposed model will be validated on more challenging CFRP specimen with irregular shape and varying depth. The proposed method will be applied across wider infrared measurement technology such as eddy current pulsed thermography (ECPT). The computational complexity of the model will be further improved for online NDT.

# VI. ACKNOWLEDGEMENT

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